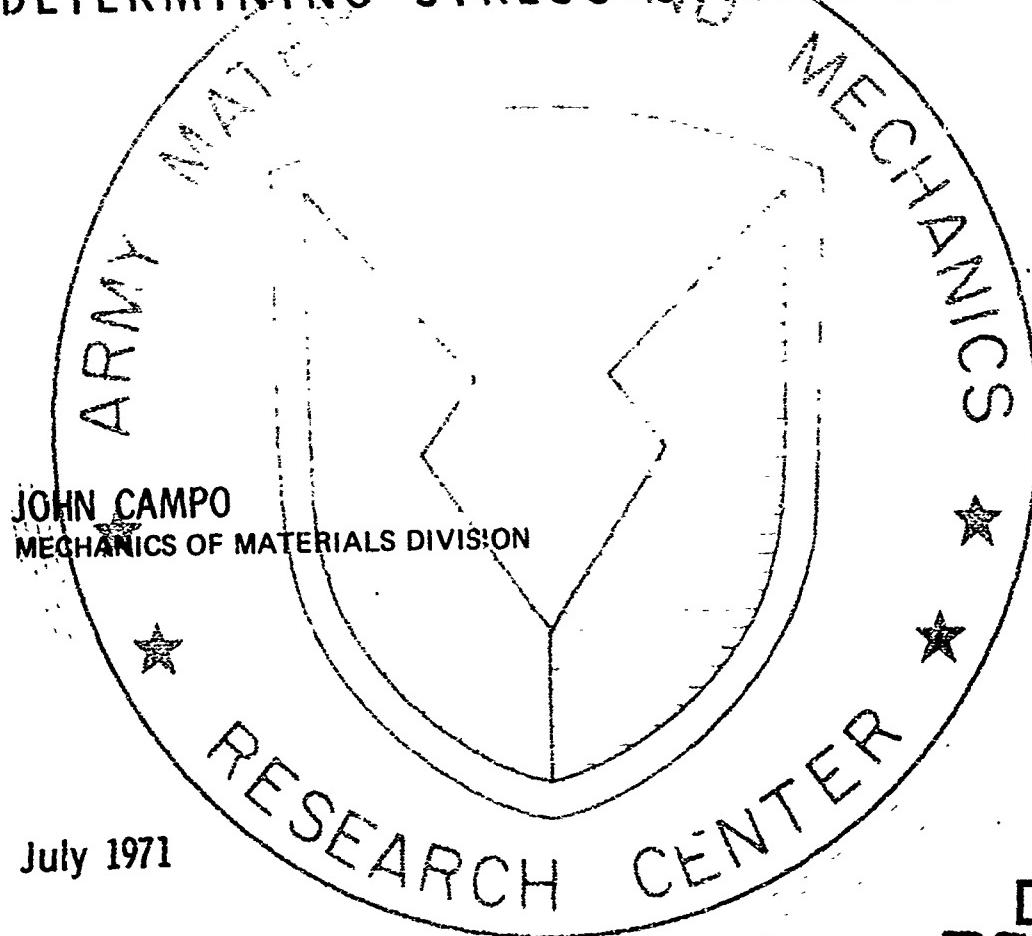


AMMRC TR 71-13

AD

AD 731203

# UTILIZATION OF THE BEND TEST FOR DETERMINING STRESS-STRAIN CURVES



July 1971

Approved for public release; distribution unlimited.

Reproduced by  
NATIONAL TECHNICAL  
INFORMATION SERVICE  
Springfield, Va. 22151

DDC  
REF ID: A65110  
OCT 20 1971  
REGULUS B

ARMY MATERIALS AND MECHANICS RESEARCH CENTER  
Watertown, Massachusetts 02172

ACCESS	
CPSTI	W. SECTION <input checked="" type="checkbox"/>
DOC	BUFF SECTION <input type="checkbox"/>
UNARMED	<input type="checkbox"/>
JUSTIFICATION	.....
BY	
DISTRIBUTION/AVAILABILITY CODES	
BIST.	AVAIL. REG/ or SPECIAL
A	

The findings in this report are not to be construed as an official Department of the Army position, unless so designated by other authorized documents.

Mention of any trade names or manufacturers in this report shall not be construed as advertising nor as an official endorsement or approval of such products or companies by the United States Government.

DISPOSITION INSTRUCTIONS

Destroy this report when it is no longer needed.  
Do not return it to the originator.

**UNCLASSIFIED**

Security Classification

**DOCUMENT CONTROL DATA - R & D**

(Security classification of title, body of abstract and indexing annotation must be entered when the overall report is classified)

1. ORIGINATING ACTIVITY (Corporate author) Army Materials and Mechanics Research Center Watertown, Massachusetts 02172		2a. REPORT SECURITY CLASSIFICATION Unclassified
2b. GROUP		
3. REPORT TITLE <b>UTILIZATION OF THE BEND TEST FOR DETERMINING STRESS-STRAIN CURVES</b>		
4. DESCRIPTIVE NOTES (Type of report and inclusive dates)		
5. AUTHOR(S) (First name, middle initial, last name) John Campo		
6. REPORT DATE July 1971	7a. TOTAL NO. OF PAGES 24	7b. NO. OF REFS 6
8a. CONTRACT OR GRANT NO.	9a. ORIGINATOR'S REPORT NUMBER(S) AMMRC TR 71-13	
b. PROJECT NO. D/A Project 1T062105A328	9b. OTHER REPORT NO(S) (Any other numbers that may be assigned this report)	
c. AMCMS Code 502E.11.294		
d. Agency Accession Number DA OB4707		
10. DISTRIBUTION STATEMENT Approved for public release; distribution unlimited.		
11. SUPPLEMENTARY NOTES	12. SPONSORING MILITARY ACTIVITY U. S. Army Materiel Command Washington, D. C. 20315	
13. ABSTRACT Experimental verification of the utilization of the bend test for determining stress-strain curves for tension and compression is presented in this report. Stress-strain curves are determined from independently performed tension, compression, and bend tests of AISI 4340 steel specimens (heat treated to three nominal yield strength levels - 125,000, 175,000, and 225,000 psi, respectively, at 0.2% offset) and compared. Fair agreement is found to exist among the curves up to the 2% strain limit of each test. (Author)		

DD FORM 1 NOV 68 1473 REPLACES DD FORM 1473, 1 JAN 64, WHICH IS  
OBsolete FOR ARMY USE.

**UNCLASSIFIED**

Security Classification

**UNCLASSIFIED**

Security Classification

14 KEY WORDS	LINK A		LINK B		LINK C	
	ROLE	WT	ROLE	WT	ROLE	WT
Bend tests Tension tests Compression tests Static tests Stress-strain diagrams Measuring instruments Strain gages						

**UNCLASSIFIED**

Security Classification

MMRC TR 71-13

**UTILIZATION OF THE BEND TEST FOR DETERMINING STRESS-STRAIN CURVES**

Technical Report by

**JOHN CAMPO**

July 1971

D/A Project 1T062105A328  
AMCMS Code 502E.11.294  
Metals Research for Army Materiel  
Agency Accession Number DA OB4707

Approved for public release; distribution unlimited.

MECHANICS OF MATERIALS DIVISION  
ARMY MATERIALS AND MECHANICS RESEARCH CENTER  
Watertown, Massachusetts 02172

## ARMY MATERIALS AND MECHANICS RESEARCH CENTER

### UTILIZATION OF THE BEND TEST FOR DETERMINING STRESS-STRAIN CURVES

#### ABSTRACT

Experimental verification of the utilization of the bend test for determining stress-strain curves for tension and compression is presented in this report. Stress-strain curves are determined from independently performed tension, compression, and bend tests of AISI 4340 steel specimens (heat treated to three nominal yield strength levels - 125,000, 175,000, and 225,000 psi, respectively, at 0.2% offset) and compared. Fair agreement is found to exist among the curves up to the 2% strain limit of each test.

## CONTENTS

	Page
ABSTRACT	
I. NOMENCLATURE . . . . .	1
II. INTRODUCTION . . . . .	1
III. VALUES OF STRESSES AND STRAINS FROM BEND TEST DATA . . . . .	2
IV. VALUES OF STRESSES AND STRAINS FROM CONVENTIONAL TENSION AND COMPRESSION TEST DATA . . . . .	3
V. SELECTION OF SPECIMEN SIZES . . . . .	4
VI. SPECIMEN PREPARATION . . . . .	4
VII. MEASURING DEVICES AND EXPERIMENTAL PROCEDURES . . . . .	5
VIII. RESULTS AND DISCUSSION . . . . .	7
IX. SUMMARY AND CONCLUSIONS . . . . .	20
ACKNOWLEDGMENTS . . . . .	20
LITERATURE CITED . . . . .	21

## I. NOMENCLATURE

### Bend Tests

$a$  = moment arm or distance between either support and its closer load point (double point loading).

$b, h$  = beam cross sectional dimensions, width and height, respectively.

$P$  = applied load.

$\sigma_1, \sigma_2$  = absolute values of extreme fiber stresses, compression and tension, respectively.

$\epsilon_1, \epsilon_2$  = absolute values of corresponding extreme fiber strains.

$\delta$  = deflection at midspan of neutral axis from its original horizontal position.

### Tension and Compression Tests

$A_0$  = original cross sectional area.

$P$  = applied load.

$\sigma = P/A_0$  = absolute value of engineering stress.

$\epsilon$  = absolute value of axial ordinary strain.

## II. INTRODUCTION

Interest in the bend test has undergone a revival in recent years because of the development and exploitation of such brittle materials as ceramics and carbides and because of the greater use of high-strength materials with little ductility.

The tension test has normally been one of the most common methods used for determining design data of a material. Sometimes, however, a tensile specimen may not be feasible either because of size limitations or because of difficulty in machining the particular material. The latter problem, usually with accompanying extraordinary expenses, is particularly encountered when dealing with such limited ductility materials as those mentioned above. In such cases, a bend test with its major advantage of employing a simple specimen with a rectangular cross section would be most welcome, provided that such a test could be used as a satisfactory substitute for the tension test.

Nadai,<sup>1</sup> in an analysis of the bending problem, provides the foundation for deriving the tensile and compressive stress-strain curves from the bend test. Bluhm<sup>2</sup> has modified the form of the resulting equation extending it to consideration of cases other than pure bending, illustrated its application, and suggested computational forms to facilitate the carrying out of the required operations.

In the present investigation, experimental verification of the proposed formulation is attempted. Tension, compression, and bend tests are independently performed on AISI 4340 steel specimens heat treated to three nominal yield-strength levels (125,000, 175,000, and 225,000 psi, respectively, at 0.2% offset). Stress-strain curves determined from the bend tests are then compared to those determined from the conventional tension and compression tests. The results tend to substantiate, at least up to the 2.0% strain limit of each test, that the proposed method is valid in obtaining a stress-strain curve for either tension or compression by utilization of the bend test.

Before continuing with the text, however, it should be brought out that: a) a steel, rather than some ceramic is chosen as the testing material in order to minimize machining and testing costs; b) an arbitrary value of 2.0% strain is selected as an upper test limit since many brittle materials will fracture below this strain value; and c) a computer program could obviously have been devised for automatic reduction of the test data.

### III. VALUES OF STRESSES AND STRAINS FROM BEND TEST DATA\*

#### Stresses

If bend tests are divided into three cases as noted below, and if, for ease in manipulation, the values of tensile and compressive stresses computed from the bend test data are expressed in absolute terms, that is, numerical values only, then the formulas presented by Bluhm become:

Case I. Different Stress-Strain Curves for Tension and Compression.

$$\sigma_1 = \left[ \frac{a}{bh^2} \right] \left[ p + \left( \frac{\epsilon_1 + \epsilon_2}{2} \right) \left( \frac{dp}{d(\epsilon_1 + \epsilon_2)} \right) \right] \left[ \frac{d\epsilon_2}{d\epsilon_1} + 1 \right] \quad (1)$$

$$\sigma_2 = \left[ \frac{a}{bh^2} \right] \left[ p + \left( \frac{\epsilon_1 + \epsilon_2}{2} \right) \left( \frac{dp}{d(\epsilon_1 + \epsilon_2)} \right) \right] \left[ \frac{1}{\frac{d\epsilon_2}{d\epsilon_1} + 1} \right] \quad (2)$$

Case II. Identical Stress-Strain Curves for Tension and Compression.

$$\sigma_1 = \sigma_2 = \left[ \frac{2a}{bh^2} \right] \left[ p + \left( \frac{\epsilon}{2} \right) \left( \frac{dp}{d\epsilon} \right) \right] \quad (3)$$

Case III. Identical Stress Strain for Tension and Compression. Beam Restricted to Pure Bending.

$$\sigma_1 = \sigma_2 = \left[ \frac{2a}{bh^2} \right] \left[ p + \left( \frac{\delta}{2} \right) \left( \frac{d\delta}{d\epsilon} \right) \right] \quad (4)$$

---

\*See Nomenclature (Section I) for definition of terms and Figure 1A for specimen configuration and loading arrangement.

## Strains

The values of strains in the bend tests are simply those indicated by the strain gages, again in absolute terms.

## Basic Consideration in Development of Formulas

Because this report is concerned chiefly with experimental verification of the expressions for determining tensile and/or compressive stress-strain curves from bend test data (Eqs. (1) through (4), inclusive), the actual derivation of the formulas, to be found in Reference 1, is intentionally omitted. However, the basic considerations involved in the development are listed here:

1. An initially straight beam of constant cross section is caused either by forces perpendicular to its longitudinal axis in one of the principal planes of inertia of the cross section or by moments in one of the planes.
2. The cross-sectional dimensions of the beam are small in comparison to its length so that shear deformations may be ignored.
3. Cross sections of the beam perpendicular to the axis remain plane during bending.
4. For a beam restricted to pure bending, it is implied that the whole beam bends into a circular arc. For the more general bending conditions, however, it is necessary that only the portion of the beam between two closely-spaced cross sections under consideration (along the length of the strain gage, for example) bend approximately into a circular arc.

## IV. VALUES OF STRESSES AND STRAINS FROM CONVENTIONAL TENSION AND COMPRESSION TEST DATA\*

### Stresses and Strains

Since the total strain in any one test does not exceed 1.0%, and since the test results are for comparison purposes only, it is deemed sufficient to determine engineering stress-ordinary strain rather than true stress-true strain curves. Consequently, the value of stress is simply that obtained by dividing the applied load by the original cross-sectional area of the specimen, and the value of strain, that obtained by averaging the values indicated by the strain gages, again in absolute terms.

\*See Nomenclature (Section I) for definition of terms and Figures 1 and 2 for specimen configuration and loading arrangements.

## V. SELECTION OF SPECIMEN SIZES

### Bend Specimen

More than one experimenter has found that varying the size of test specimens leads to different bend test properties. From the data presented by Grobe and Roberts<sup>3</sup> (when employing the double-point loading system) it may be seen that higher bend-strength values are obtained from a 2-in. rather than a 4-in. span (distance between outermost supports) or from a 0.250-in. rather than a 0.500-in. beam width. Burghardt and Elder<sup>4</sup> remind us to maintain a reasonable span (they chose 0.500-in.) and keep the h/a ratio, that is, ratio of beam height to moment arm\*, smaller than 2/3 in order to prevent shear stresses from becoming too pronounced. Duckworth<sup>5</sup> also suggests that the h/a ratio be kept as small as practicable to avoid pronounced shear stresses, and, in addition, that the b/h ratio, that is, ratio of beam width to beam height, be kept reasonably small to avoid stiffening effect due to relatively wide beams. Because of the above findings and because of space requirements for the application of strain gages and their leads, a 3-in. span was judged to be satisfactory, and a bar with a 3 1/4-in. overall length, a 0.600-in. width, and a 0.150-in. height was chosen as the bend specimen.

### Tension Specimen

A threaded tensile specimen with a 0.357-in. uniform diameter was used in this investigation.

### Compression Specimen

According to ASTM EN-61,<sup>6</sup> a compression specimen should be round, and, if of medium length, should have a length-to-diameter ratio of 3.0. A 5/8-in. diameter was selected in order to keep the applied load from exceeding the 120,000 pounds capacity of the testing machine and, yet, to have ample space along the specimen for strain-gage application. In compliance with the recommended length-to-diameter ratio of 3.0, the specimen was made 2.0 in. long.

## VI. SPECIMEN PREPARATION

### Material

AISI 4340 steel was chosen for the experimental work in this investigation. In order to insure uniform material properties, all specimens were taken from one plate, 12 x 12 x 3/4 in., and the length of each specimen was always in the longitudinal direction of the plate. In addition, the width of each bend specimen was taken in the transverse direction of the plate.

---

\*See Figure 1A for explanation of moment arm, a.

## Heat Treatment

Roughly cut blanks, all with the original 3/4-in. thickness were heat treated before machining, as indicated below, to three nominal yield strength values (125,000, 175,000, and 225,000 psi, respectively, at 0.2% offset). All blanks were normalized by heating at 1650°F for one hr and by air cooling. They were then austenitized at 1550°F for one hr and oil quenched. Double tempering followed at the temperatures and for the periods of time indicated:

Level 1 - nominal yield strength of 125,000 psi - 1200°F (1+1) hr.

Level 2 - nominal yield strength of 175,000 psi - 950°F (1/2+1/2) hr.

Level 3 - nominal yield strength of 225,000 psi - 400°F (2+2) hr.

Tempering was performed in salt baths.

## Surface Grinding

After heat treatment, the bend specimens were brought to within 0.010 or 0.015 in. of the final dimensions by rough grinding and were then finished by fine surface grinding. A similar procedure was followed for the ends of the compression specimens.

## Number of Specimens

Four tension, 4 compression, and 5 bend specimens in each of the three nominal yield-strength levels, or a total of 39 specimens in all, were tested.

# VII. MEASURING DEVICES AND EXPERIMENTAL PROCEDURES

## Bend Tests

Since strain gage readings on both the top and bottom surfaces were required, double-point loading (see Figure 1A) was desirable. Longitudinal strain gages (SR4, type FAE-25-PL) were mounted, one each of mid-span on both the top and bottom surfaces of each specimen. The strain gages were electrically connected to separate X-Y recorders so that simultaneous readings could be made.

The testing was performed on a Tinius Olsen testing machine, manually operated at head speeds well within the limits recommended in ASTM standards.<sup>6</sup> Strain-rate effects were considered to be negligible. A signal from a pressure transducer in the testing machine, indicating the applied load, was also connected to the X-Y recorders, and, in this way, plots of load vs strain were autographically recorded. Each test was continued until a total strain of 2.0% was indicated. Figure 2 shows the test setup.

In this setup, load and support points consisted of "rollers" that were free to rotate in order to minimize friction effects.

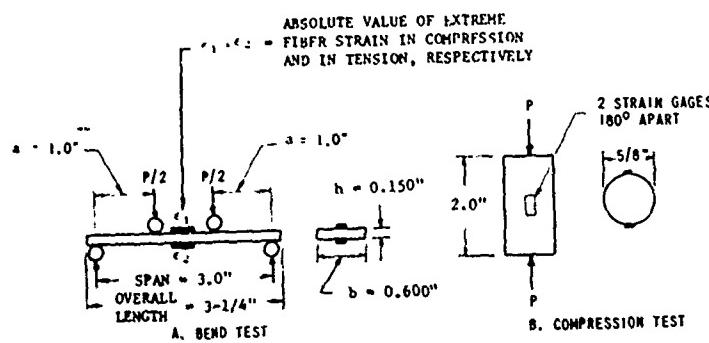
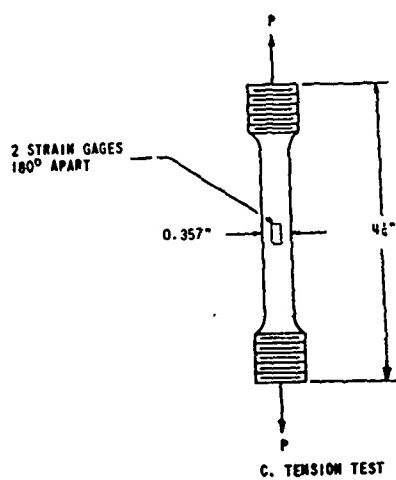


Figure 1. Sketch of loading arrangement for bend, compression, and tension test specimens



19-066-1250/AMC-68

Figure 2. Bend test setup (A), employing "friction free" fixture (B), showing specimen under load. Double point loading.  
0.150 x 0.600 x 3.25-inch specimen

## Tension and Compression Tests

Since axial ordinary strains of the specimens were required in these tests, longitudinal strain gages (SR4, type PFA-50-12-L) were mounted at midspan and  $180^\circ$  apart (see Figures 1B and 1C). These gages were connected to an X-Y recorder so that average strain only could be recorded.

The testing was also performed on a Tinius Olsen testing machine, and, again, the head speeds were kept well within the limits recommended in ASTM standards. The load signal of the testing machine was also connected to the X-Y recorder, but the recorder was calibrated so that engineering stress (rather than applied load) could be indicated directly. For the tension and compression tests, then, plots of engineering stress vs strain were autographically recorded. Again, each test was continued until a total strain of 2.0% was reached. Figure 3 illustrates the testing arrangement for tension tests and Figure 4 that for compression tests.

## VIII. RESULTS AND DISCUSSION

### Recorded Data

Results were generally obtained in the form of X-Y recordings. These recordings consisted of applied load vs extreme fiber strains for the bend tests and engineering stress-strain curves, directly, for the tensile and compressive tests.

### Auxiliary Curves for Bend Tests

When the extreme fiber strains of a bend test specimen were not equal, two auxiliary curves had to be constructed before any stress-strain curve could be determined. These auxiliary curves consisted of two plots, one of applied load vs sum of extreme fiber strains and one of extreme fiber strain in compression vs extreme fiber strain in tension. No such auxiliary curve was required when the extreme fiber strains were equal.

### Stress-Strain Curves for Each Specimen

The originally obtained X-Y recordings in the tensile and compressive tests consisted of engineering stress-strain curves, directly, and no further treatment of the data was required in determining these results. For the bend tests, however, to determine the required stress-strain curves, Eqs. (1) and (2) of the text along with slope data of the auxiliary curves were employed when the extreme fiber strains were not equal, and Eq. (3) of the text along with slope data of either of the originally obtained X-Y recordings were employed when the extreme fiber strains were equal.

### Stress-Strain Curves for Each Yield Strength Level

In addition, because the stress-strain curves were essentially identical for any one strength level and for any one type of testing, stress-strain curves for each yield strength level were determined by averaging the individual test

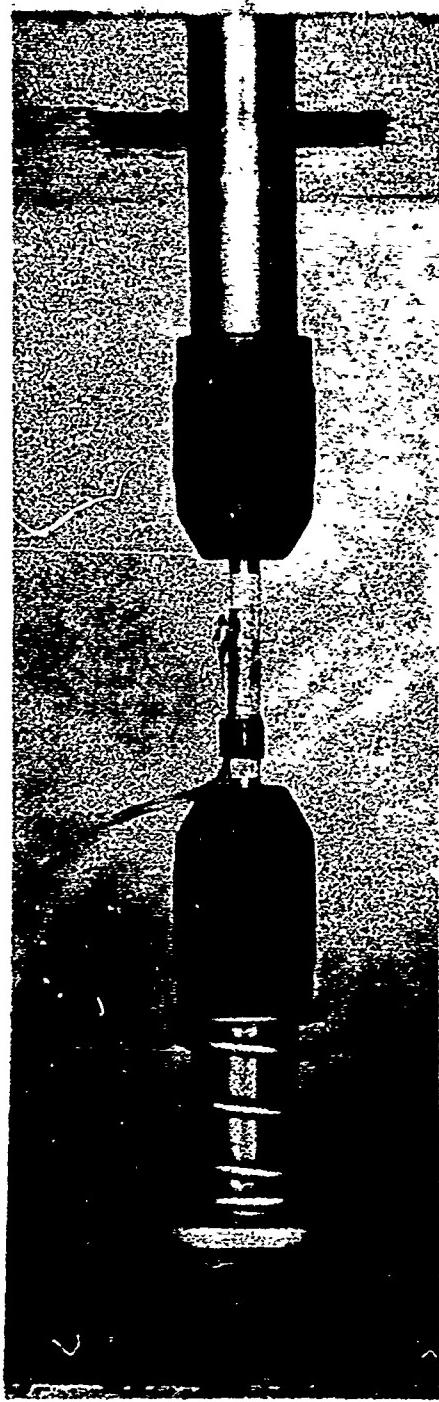


Figure 3. Tension test setup showing specimen under load. 0.357-inch uniform diameter x 4.25-inch threaded type specimen

19-066-1376/AMC-69

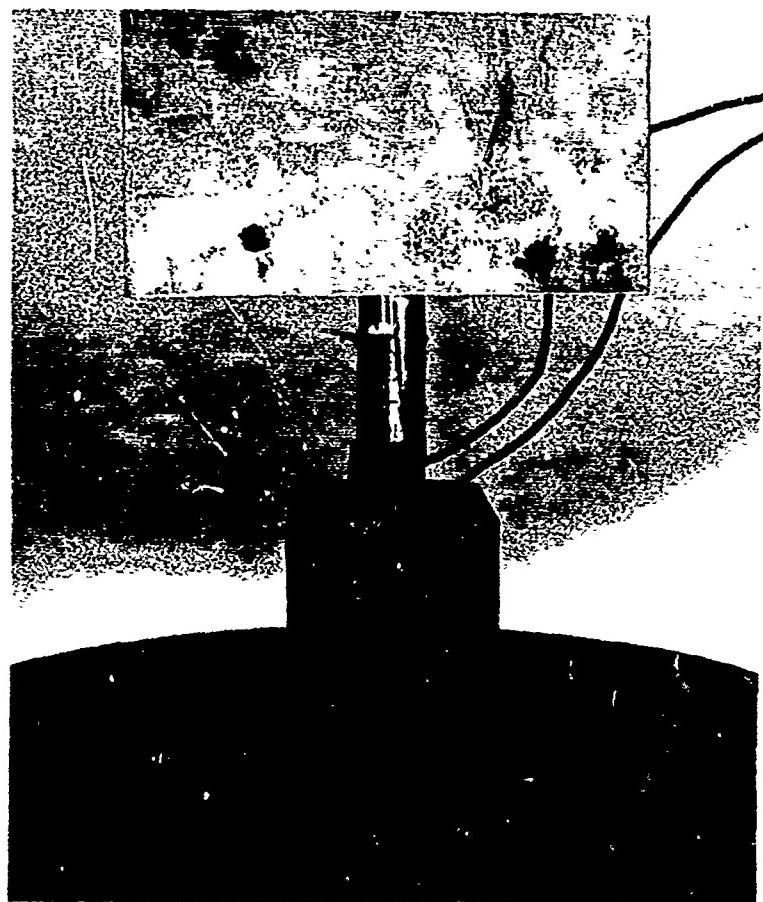


Figure 4. Compression test setup showing specimen under load.  
0.625-inch diameter x 2.0-inch specimen  
19-066-1377/AMC-69

results. It was felt that a comparison of the stress-strain curves based on the average test results would be less bulky and just as indicative as a comparison of the curves based on the individual test results themselves.

### Reported Results

The average test data and corresponding values of stresses and strains for all three nominal yield-strength level specimens are summarized in Tables I through IV. Stress-strain curves are given in Figure 5 where the plotted values are taken from the aforementioned tables. In addition, pertinent values of mechanical properties (including values of Young's moduli, yield strength at 0.2% offset and flow stresses at both 1.0% and 2.0% of total strain - all determined from the stress-strain curves of Figure 5) are listed in Tables V through VII. Maximum percentage scatter and percentage discrepancies (both these terms are subsequently defined and discussed) are also included in these tables.

Auxiliary curves, when required, and stress-strain curves, both in tension and in compression, have been constructed for each bend specimen as well as for each yield strength level. As typical examples, however, only those curves for yield strength level 1 are presented in Figures 6, 7, and 8. In fact, only the stress-strain curve in compression is depicted in the last figure.

### Scatter

If scatter is defined as the amount that an individual test value differs from the average test value of a particular number of specimens tested, then

$$\% \text{ scatter} = \frac{\text{individual test value} - \text{average test value}}{\text{average test value}} \times 100.$$

Scatter is expected to be greater in stress-strain curves derived from bend tests than those derived from conventional tests. This is true because, as described above, in conventional tests, stress-strain curves may be obtained directly, whereas, in bend tests, slope data must first be determined either from one of the originally obtained X-Y recordings or from auxiliary constructed curves. The work involved in obtaining the required slope data introduces more leeway for experimental error and, hence, greater scatter.

### Discrepancy

If discrepancy is defined as the amount that a bend test value differs from the corresponding conventional test value, then

$$\% \text{ discrepancy} = \frac{\text{bend test value} - \text{conventional test value}}{\text{conventional test value}} \times 100.$$

Discrepancy values, then, indicate the degree of agreement, or rather, of disagreement, among results predicted by bend tests and those determined by conventional tests.

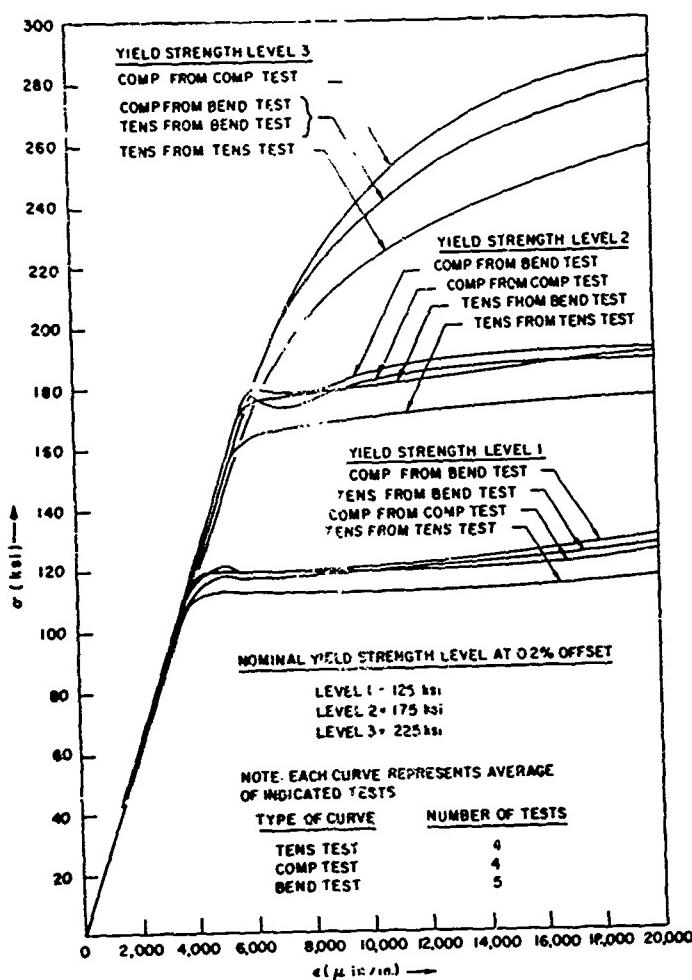


Figure 5. Stress-strain curves from bend, tension, and compression tests

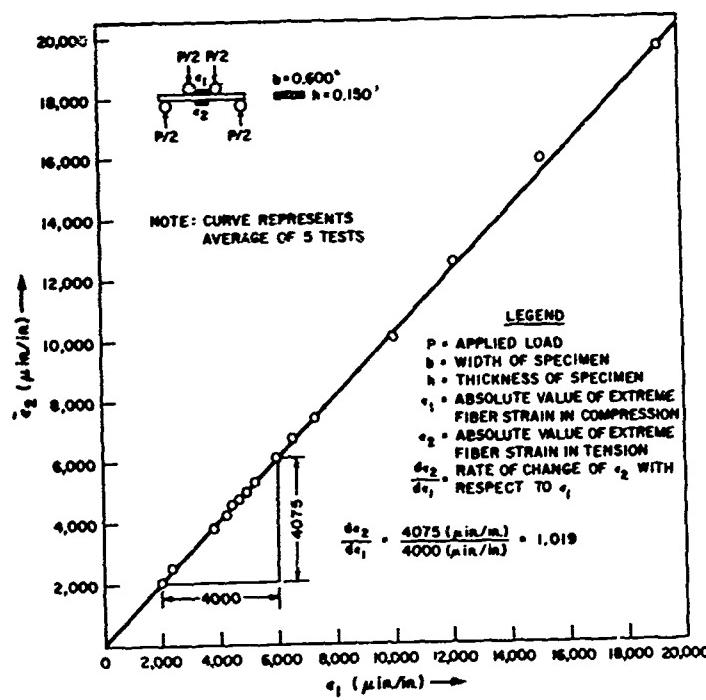


Figure 6. Plot of tensile strain vs compressive strain for yield strength level 1 bend specimens (auxiliary curve)

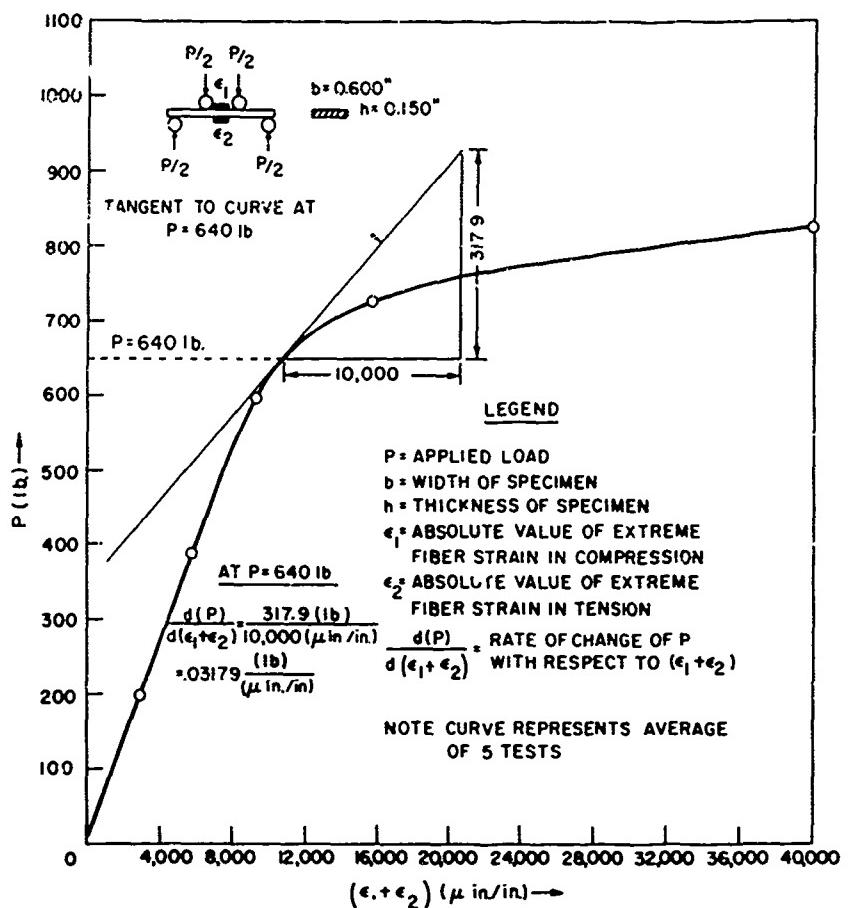


Figure 7. Plot of load vs sum of strains for yield strength level 1 bend specimens (auxiliary curve)

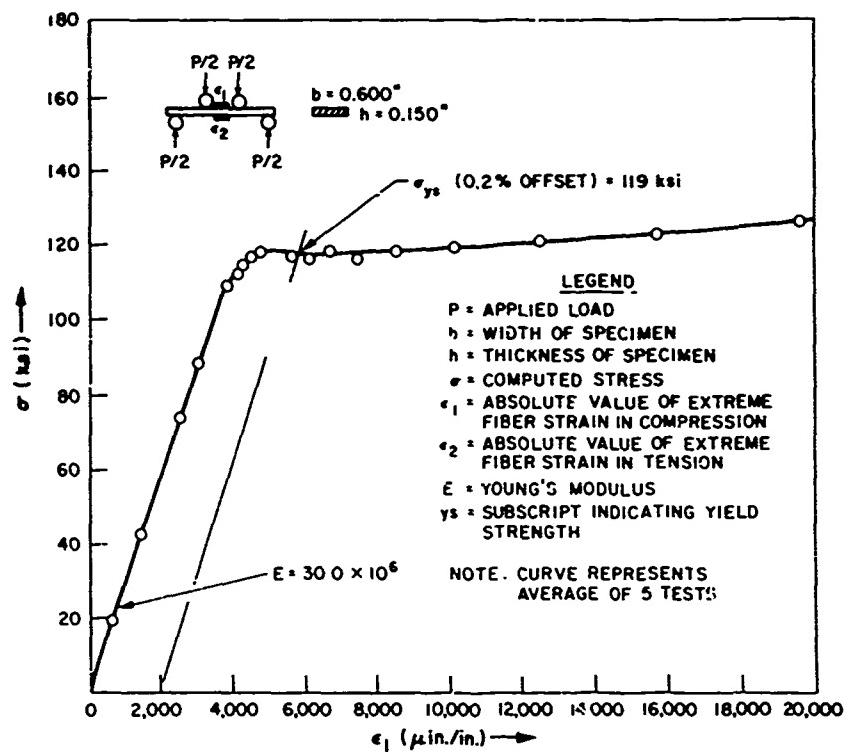


Figure 8. Plot of stress vs strain in compression for yield strength level 1 bend specimens

Table I. AVERAGE STRESS-STRAIN VALUES FROM  
YIELD-STRENGTH LEVEL 1 BEND SPECIMENS

(1) $P$ (lb)	(2) $\epsilon_1$	(3) $\epsilon_2$	(4) $\frac{\epsilon_1 + \epsilon_2}{2}$	(5) $\frac{dP}{d(\epsilon_1 + \epsilon_2)}$ (lb/ $\mu\text{in/in}$ )	(6) $\sigma_1$ (lb/ $\text{in}^2$ )	(7) $\sigma_2$ (lb/ $\text{in}^2$ )
0	0	0	0	0.06614	0	0
100	749	763	756	0.06614	22,433	22,015
200	1498	1526	1512	0.06614	44,866	44,030
300	2247	2289	2268	0.06614	67,300	66,045
400	2996	3052	3024	0.06614	89,733	88,060
500	3745	3815	3780	0.06614	112,167	110,075
520	3881	3953	3917	0.06063	113,287	111,174
540	4055	4131	4093	0.05423	113,957	111,832
560	4240	4319	4280	0.05107	116,441	114,270
580	4433	4515	4474	0.04850	119,194	116,972
600	4657	4745	4701	0.04325	120,141	117,901
620	4927	5019	4973	0.03800	120,986	118,730
640	5196	5294	5254	0.03179	120,652	118,403
660	5553	5557	5605	0.02370	118,574	116,363
680	6004	6116	6060	0.01808	118,083	115,881
700	6608	6732	6670	0.01550	120,151	117,911
720	7292	7428	7360	0.00962	118,268	116,063
740	8372	8528	8450	0.00732	119,897	117,662
760	9947	10,133	10,040	0.00507	121,274	119,013
780	12,242	12,470	12,356	0.00392	123,898	121,159
800	15,406	15,694	15,550	0.00287	126,319	123,964
820	19,260	19,620	19,440	0.00245	129,759	127,339

NOTES:

1) Circled numbers identify appropriate columns

2) 
$$(6) = \left[ \frac{a}{bh^2} \right] \left[ (1) + (4) \times (5) \right] \left[ \frac{d\epsilon_2}{d\epsilon_1} + 1 \right] \quad \text{Eq. (1)}$$

3) 
$$(7) = \left[ \frac{a}{bh^2} \right] \left[ (1) + (4) \times (5) \right] \left[ \frac{1}{d\epsilon_2} + 1 \right] \quad \text{Eq. (2)}$$

4) 
$$\frac{a}{bh^2} = \frac{1.0}{(.600)(.150^2)} = 74.074$$

5) 
$$\frac{d\epsilon_2}{d\epsilon_1} = 1.019$$

6) Above values represent averages of five tests

Table II. AVERAGE STRESS-STRAIN VALUES FROM  
YIELD-STRENGTH LEVEL 2 BEND SPECIMENS

(1) P (lb)	(2) $\epsilon_1$ --	(3) $\epsilon_2$ ( $\mu\text{in/in}$ )	(4) $\frac{\epsilon_1 + \epsilon_2}{2}$ --	(5) $\frac{dP}{d(\epsilon_1 + \epsilon_2)}$ ( $\text{lb}/\mu\text{in/in}$ )	(6) $\sigma_1$ ( $\text{lb}/\text{in}^2$ )	(7) $\sigma_2$ ( $\text{lb}/\text{in}^2$ )
0	0	0	0	0.06717	0	0
200	1478	1500	1489	0.06717	44,778	44,116
400	2955	3000	2978	0.06717	89,555	88,232
600	4433	4500	4467	0.06717	134,333	132,348
800	5911	5999	5955	0.06717	179,111	176,464
820	6052	6142	6097	0.06098	177,886	175,257
840	6213	6306	6260	0.05917	180,663	177,993
860	6396	6492	6444	0.05181	178,194	175,561
880	6604	6703	6654	0.04608	177,114	174,496
900	6898	7002	6950	0.03788	173,629	171,063
920	7084	7191	7138	0.03623	175,918	173,319
940	7520	7430	7375	0.03367	177,368	174,746
960	7668	7783	7726	0.02494	172,049	169,507
980	8015	8135	8075	0.02778	179,756	177,099
1000	8402	8528	8465	0.02370	179,203	176,555
1020	8769	8901	8835	0.02110	180,069	177,408
1040	9281	9420	9351	0.01930	182,166	179,474
1060	9851	9999	9925	0.01720	183,695	180,980
1080	10,484	10,642	10,563	0.01475	184,454	181,728
1100	11,232	11,400	11,316	0.01330	186,649	183,890
1120	12,010	12,190	12,100	0.01180	188,481	185,696
1140	12,854	13,046	12,950	0.00975	189,001	186,208
1160	13,995	14,205	14,100	0.00780	189,556	186,755
1180	15,533	15,767	15,650	0.00580	189,674	186,871
1200	17,866	18,134	18,000	0.00400	189,858	187,052
1210	18,821	19,104	18,763	0.00330	189,944	187,137
1220	20,261	20,564	20,413	0.00260	190,017	187,209

NOTES:

1) Circled numbers identify appropriate columns

$$2) (6) = \left[ \frac{a}{bh^2} \right] \left[ (1) + (4) \times (5) \right] \left[ \frac{d\epsilon_2}{d\epsilon_1} + 1 \right] \quad \text{Eq. (1)}$$

$$3) (7) = \left[ \frac{a}{bh^2} \right] \left[ (1) + (4) \times (5) \right] \left[ 1 / \frac{d\epsilon_2}{d\epsilon_1} + 1 \right] \quad \text{Eq. (2)}$$

$$4) \frac{a}{bh^2} = \frac{1.0}{(.600)(.150)^2} = 74.074 \quad 5) \frac{d\epsilon_2}{d\epsilon_1} = 1.015$$

6) Above values represent averages of five tests

Table III. AVERAGE STRESS-STRAIN VALUES FROM  
YIELD-STRENGTH LEVEL 3 BEND SPECIMENS

(1) $P$ (lb)	(2) $\epsilon_1 = \epsilon_2 = \epsilon$ (in/in)	(3) $\frac{dp}{d\epsilon}$ (lb/in/in)	(4) $\sigma_1 = \sigma_2$ (lb/in <sup>2</sup> )
0	0	0.13098	0
200	1527	0.13098	44,444
400	3054	0.13098	58,533
600	4581	0.13098	73,333
800	6108	0.13098	87,778
840	6414	0.12880	105,639
880	6734	0.12640	103,420
920	7055	0.12120	109,634
960	7402	0.11000	202,535
1000	7769	0.10440	205,228
1040	8171	0.10080	215,084
1080	8554	0.09470	220,005
1120	8991	0.08960	225,591
1160	9446	0.08060	228,248
1200	9942	0.07600	233,747
1240	10,490	0.07250	240,558
1280	11,068	0.06950	246,698
1320	11,607	0.06100	245,002
1360	12,358	0.05560	232,373
1400	13,114	0.04830	254,902
1440	13,948	0.04450	253,510
1480	14,936	0.04050	264,067
1520	15,875	0.03620	233,459
1560	16,992	0.03240	271,832
1580	17,542	0.03050	273,446
1600	18,164	0.02850	273,114
1620	19,054	0.02640	277,222
1640	19,792	0.02460	279,028

NOTES:

1) Circled numbers identify appropriate columns

2) 
$$\textcircled{4} = \left[ \frac{2a}{3h^2} \right] \left[ \textcircled{1} + \frac{\textcircled{2} \times \textcircled{3}}{2} \right] \quad \text{Eq. (5)}$$

3) 
$$\frac{2a}{3h^2} = \frac{(3)(1.0)}{(0.600)(0.150)^2} = 148.148$$

4) Above values represent averages of five tests

如图 11-4 所示，当风速增加时，风压会减小，从而产生负压，使气流从低处向高处流动。

Row	Line	Column	Value
1	1	1	1
2	2	1	1
3	3	1	1
4	4	1	1
5	5	1	1
6	6	1	1
7	7	1	1
8	8	1	1
9	9	1	1
10	10	1	1
11	11	1	1
12	12	1	1
13	13	1	1
14	14	1	1
15	15	1	1
16	16	1	1
17	17	1	1
18	18	1	1
19	19	1	1
20	20	1	1
21	21	1	1
22	22	1	1
23	23	1	1
24	24	1	1
25	25	1	1
26	26	1	1
27	27	1	1
28	28	1	1
29	29	1	1
30	30	1	1
31	31	1	1
32	32	1	1
33	33	1	1
34	34	1	1
35	35	1	1
36	36	1	1
37	37	1	1
38	38	1	1
39	39	1	1
40	40	1	1
41	41	1	1
42	42	1	1
43	43	1	1
44	44	1	1
45	45	1	1
46	46	1	1
47	47	1	1
48	48	1	1
49	49	1	1
50	50	1	1
51	51	1	1
52	52	1	1
53	53	1	1
54	54	1	1
55	55	1	1
56	56	1	1
57	57	1	1
58	58	1	1
59	59	1	1
60	60	1	1
61	61	1	1
62	62	1	1
63	63	1	1
64	64	1	1
65	65	1	1
66	66	1	1
67	67	1	1
68	68	1	1
69	69	1	1
70	70	1	1
71	71	1	1
72	72	1	1
73	73	1	1
74	74	1	1
75	75	1	1
76	76	1	1
77	77	1	1
78	78	1	1
79	79	1	1
80	80	1	1
81	81	1	1
82	82	1	1
83	83	1	1
84	84	1	1
85	85	1	1
86	86	1	1
87	87	1	1
88	88	1	1
89	89	1	1
90	90	1	1
91	91	1	1
92	92	1	1
93	93	1	1
94	94	1	1
95	95	1	1
96	96	1	1
97	97	1	1
98	98	1	1
99	99	1	1
100	100	1	1
101	101	1	1
102	102	1	1
103	103	1	1
104	104	1	1
105	105	1	1
106	106	1	1
107	107	1	1
108	108	1	1
109	109	1	1
110	110	1	1
111	111	1	1
112	112	1	1
113	113	1	1
114	114	1	1
115	115	1	1
116	116	1	1
117	117	1	1
118	118	1	1
119	119	1	1
120	120	1	1
121	121	1	1
122	122	1	1
123	123	1	1
124	124	1	1
125	125	1	1
126	126	1	1
127	127	1	1
128	128	1	1
129	129	1	1
130	130	1	1
131	131	1	1
132	132	1	1
133	133	1	1
134	134	1	1
135	135	1	1
136	136	1	1
137	137	1	1
138	138	1	1
139	139	1	1
140	140	1	1
141	141	1	1
142	142	1	1
143	143	1	1
144	144	1	1
145	145	1	1
146	146	1	1
147	147	1	1
148	148	1	1
149	149	1	1
150	150	1	1
151	151	1	1
152	152	1	1
153	153	1	1
154	154	1	1
155	155	1	1
156	156	1	1
157	157	1	1
158	158	1	1
159	159	1	1
160	160	1	1
161	161	1	1
162	162	1	1
163	163	1	1
164	164	1	1
165	165	1	1
166	166	1	1
167	167	1	1
168	168	1	1
169	169	1	1
170	170	1	1
171	171	1	1
172	172	1	1
173	173	1	1
174	174	1	1
175	175	1	1
176	176	1	1
177	177	1	1
178	178	1	1
179	179	1	1
180	180	1	1
181	181	1	1
182	182	1	1
183	183	1	1
184	184	1	1
185	185	1	1
186	186	1	1
187	187	1	1
188	188	1	1
189	189	1	1
190	190	1	1
191	191	1	1
192	192	1	1
193	193	1	1
194	194	1	1
195	195	1	1
196	196	1	1
197	197	1	1
198	198	1	1
199	199	1	1
200	200	1	1
201	201	1	1
202	202	1	1
203	203	1	1
204	204	1	1
205	205	1	1
206	206	1	1
207	207	1	1
208	208	1	1
209	209	1	1
210	210	1	1
211	211	1	1
212	212	1	1
213	213	1	1
214	214	1	1
215	215	1	1
216	216	1	1
217	217	1	1
218	218	1	1
219	219	1	1
220	220	1	1
221	221	1	1
222	222	1	1
223	223	1	1
224	224	1	1
225	225	1	1
226	226	1	1
227	227	1	1
228	228	1	1
229	229	1	1
230	230	1	1
231	231	1	1
232	232	1	1
233	233	1	1
234	234	1	1
235	235	1	1
236	236	1	1
237	237	1	1
238	238	1	1
239	239	1	1
240	240	1	1
241	241	1	1
242	242	1	1
243	243	1	1
244	244	1	1
245	245	1	1
246	246	1	1
247	247	1	1
248	248	1	1
249	249	1	1
250	250	1	1
251	251	1	1
252	252	1	1
253	253	1	1
254	254	1	1
255	255	1	1
256	256	1	1
257	257	1	1
258	258	1	1
259	259	1	1
260	260	1	1
261	261	1	1
262	262	1	1
263	263	1	1
264	264	1	1
265	265	1	1
266	266	1	1
267	267	1	1
268	268	1	1
269	269	1	1
270	270	1	1
271	271	1	1
272	272	1	1
273	273	1	1
274	274	1	1
275	275	1	1
276	276	1	1
277	277	1	1
278	278	1	1
279	279	1	1
280	280	1	1
281	281	1	1
282	282	1	1
283	283	1	1
284	284	1	1
285	285	1	1
286	286	1	1
287	287	1	1
288	288	1	1
289	289	1	1
290	290	1	1
291	291	1	1
292	292	1	1
293	293	1	1
294	294	1	1
295	295	1	1
296	296	1	1
297	297	1	1
298	298	1	1
299	299	1	1
300	300	1	1
301	301	1	1
302	302	1	1
303	303	1	1
304	304	1	1
305	305	1	1
306	306	1	1
307	307	1	1
308	308	1	1
309	309	1	1
310	310	1	1
311	311	1	1
312	312	1	1
313	313	1	1
314	314	1	1
315	315	1	1
316	316	1	1
317	317	1	1
318	318	1	1
319	319	1	1
320	320	1	1
321	321	1	1
322	322	1	1
323	323	1	1
324	324	1	1
325	325	1	1
326	326	1	1
327	327	1	1
328	328	1	1
329	329	1	1
330	330	1	1
331	331	1	1
332	332	1	1
333	333	1	1
334	334	1	1
335	335	1	1
336	336	1	1
337	337	1	1
338	338	1	1
339	339	1	1
340	340	1	1
341	341	1	1
342	342	1	1
343	343	1	1
344	344	1	1
345	345	1	1
346	346	1	1
347	347	1	1
348	348	1	1
349	349	1	1
350	350	1	1
351	351	1	1
352	352	1	1
353	353	1	1
354	354	1	1
355	355	1	1
356	356	1	1
357	357	1	1
358	358	1	1
359	359	1	1
360	360	1	1
361	361	1	1
362	362	1	1
363	363	1	1
364	364	1	1
365	365	1	1
366	366	1	1
367	367	1	1
368	368	1	1
369	369	1	1
370	370	1	1
371	371	1	1
372	372	1	1
373	373	1	1
374	374	1	1
375	375	1	1
376	376	1	1
377	377	1	1
378	378	1	1
379	379	1	1
380	380	1	1
381	381	1	1
382	382	1	1
383	383	1	1
384	384	1	1
385	385	1	1
3			

卷之三

- Engineering factors • Design Material Environment Manufacturing Assembly
  - Social factors • Human Society Environment
  - Some recent development strategies of four types • Design Material Manufacturing Assembly at each of the four concepts

Table V. PERTINENT MECHANICAL PROPERTIES OF YIELD-STRENGTH LEVEL 1 SPECIMENS  
INCLUDING MAXIMUM PERCENTAGE SCATTER AND PERCENTAGE DISCREPANCIES

Property in Tension	Determined from Tension Test (psi)	Maximum Scatter (%)	Determined from Bend Test (psi)	Maximum Scatter (%)	Discrepancy (%)
Young's Modulus	$29.4 \times 10^6$	+0.3 -0.2	$28.9 \times 10^6$	+0.7 -2.6	-1.7
Yield Strength at 0.2% Offset	112,750	+1.1 -0.7	117,000	+1.7 -2.6	+3.8
Flow Stress at 1.0% Strain	112,750	+1.5 -0.3	118,900	+0.5 -2.0	+5.5
Flow Stress at 2.0% Strain	116,000	0	127,400	+0.5 -1.1	+9.8

Property in Compression	Determined from Compression Test (psi)	Maximum Scatter (%)	Determined from Bend Test (psi)	Maximum Scatter (%)	Discrepancy (%)
Young's Modulus	$30.0 \times 10^6$	+1.1 -2.2	$30.0 \times 10^6$	+0.9 -3.4	0
Yield Strength at 0.2% Offset	119,000	+1.4 -0.8	119,000	+0.4 -1.7	0
Flow Stress at 1.0% Strain	119,500	+1.0 -0.9	121,300	+0.2 -2.7	+1.5
Flow Stress at 2.0% Strain	125,100	+1.0 -0.9	130,200	+0.6 -3.2	+4.1

NOTES:

1) % Scatter =  $\frac{\text{Individual Test Value} - \text{Average Test Value}}{\text{Average Test Value}} \times 100$

2) % Discrepancy =  $\frac{\text{Bend Test Value} - \text{Conventional Test Value}}{\text{Conventional Test Value}} \times 100$

3) Above values represent averages of four tests in tension, four tests in compression, and five tests in bending

Table VI. PERTINENT MECHANICAL PROPERTIES OF YIELD-STRENGTH LEVEL 2 SPECIMENS  
INCLUDING MAXIMUM PERCENTAGE SCATTER AND PERCENTAGE DISCREPANCIES

Property in Tension	Determined from Tension Test (psi)	Maximum Scatter (%)	Determined from Bend Test (psi)	Maximum Scatter (%)	Discrepancy (%)
Young's Modulus	$29.1 \times 10^6$	+1.5 -0.2	$29.4 \times 10^6$	+0.3 -0.5	+1.0
Yield Strength at 0.2% Offset	166,000	+0.9 -0.1	175,000	+3.9 -1.1	+5.4
Flow Stress at 1.0% Strain	169,900	+0.5 -0.2	181,000	+1.1 -2.3	+6.5
Flow Stress at 2.0% Strain	175,600	+1.4 -0.9	187,200	+2.0 -0.2	+6.6

Property in Compression	Determined from Compression Test (psi)	Maximum Scatter (%)	Determined from Bend Test (psi)	Maximum Scatter (%)	Discrepancy (%)
Young's Modulus	$30.4 \times 10^6$	+1.5 -1.2	$30.3 \times 10^6$	+2.0 -3.5	-0.3
Yield Strength at 0.2% Offset	177,000	+0.4 -1.0	178,000	+0.7 -1.1	+0.6
Flow Stress at 1.0% Strain	179,900	+0.6 -1.1	183,800	+0.2 -2.1	+2.2
Flow Stress at 2.0% Strain	188,600	+0.6 -1.0	190,000	+2.1 -0.2	+0.7

NOTES:

1) % Scatter =  $\frac{\text{Individual Test Value} - \text{Average Test Value}}{\text{Average Test Value}} \times 100$

2) % Discrepancy =  $\frac{\text{Bend Test Value} - \text{Conventional Test Value}}{\text{Conventional Test Value}} \times 100$

3) Above values represent averages of four tests in tension, four tests in compression, and five tests in bending

Table VII. PERTINENT MECHANICAL PROPERTIES OF YIELD-STRENGTH LEVEL 3 SPECIMENS  
INCLUDING MAXIMUM PERCENTAGE SCATTER AND PERCENTAGE DISCREPANCIES

Property in Tension	Determined from Tension Test (psi)	Maximum Scatter (%)	Determined from Bend Test (psi)	Maximum Scatter (%)	Discrepancy (%)
Young's Modulus	$28.9 \times 10^6$	+1.5 -1.4	$29.1 \times 10^6$	+0.9 -3.1	+0.7
Yield Strength at 0.2% Offset	214,000	+0.6 -0.8	232,000	+3.0 -1.5	+8.4
Flow Stress at 1.0% Strain	218,500	+1.1 -0.7	234,500	+1.9 -1.6	+7.3
Flow Stress at 2.0% Strain	257,350	+1.0 -1.5	279,500	+1.5 -1.8	+8.6

Property in Compression	Determined from Compression Test (psi)	Maximum Scatter (%)	Determined from Bend Test (psi)	Maximum Scatter (%)	Discrepancy (%)
Young's Modulus	$29.8 \times 10^6$	+1.3 -0.2	$29.1 \times 10^6$	+1.8 -3.8	-2.4
Yield Strength at 0.2% Offset	242,000	+0.5 -1.0	232,000	+1.0 -2.4	-4.1
Flow Stress at 1.0% Strain	240,750	+0.7 -0.6	234,500	+0.9 -1.7	-2.6
Flow Stress at 2.0% Strain	287,200	+0.6 -0.8	279,500	+1.4 -1.6	-2.7

NOTES:

- 1) % Scatter =  $\frac{\text{Individual Test Value} - \text{Average Test Value}}{\text{Average Test Value}} \times 100$
- 2) % Discrepancy =  $\frac{\text{Bend Test Value} - \text{Conventional Test Value}}{\text{Conventional Test Value}} \times 100$
- 3) Above values represent averages of four tests in tension, four tests in compression, and five tests in bending

### Acceptable Limits of Maximum Scatter

An experimental error analysis was made based on errors of  $\pm 1.0\%$  in strain gage indications,  $\pm 0.5\%$  in load indications,  $\pm 0.001$  in. in cross-sectional dimensions, and  $\pm 0.025$  in. in X-Y chart readings. According to this analysis, experimental errors could be as high as  $\pm 1.5\%$  in conventional tests and  $\pm 4.0\%$  in bend tests. These values of maximum expected experimental errors, then, are the acceptable limits of maximum scatter.

### Comparison of Stress-Strain Curves

From Figure 5, it may be seen that stress-strain curves determined from bend tests agree fairly well with those determined from conventional tests. Also from Figure 5, it may be noted from the conventional tests that, at any strain level, the corresponding values of stress in compression are generally higher than those in tension. This same trend is brought out by the bend tests.

Observations concerning both scatter and discrepancies are best noted from the results reported in Tables V through VII.

Scatter, as expected, was generally higher in bend tests than in conventional tests. The acceptable limits of maximum-percentage scatter were never exceeded in the bend tests and only once in the conventional tests — in the determination of Young's modulus in compression for yield strength level 1 specimens. Even in this case, however, scatter was considered acceptable not only because of the low value of scatter determined (-2.2%) but also because of the limited number of tests involved (four). Maximum percentage scatter, then, was generally within the computed allowable limits, and, therefore scatter was not deemed to be objectionable in any of the testing.

Discrepancy values, as mentioned earlier, are measures of agreement among results predicted by the bend tests and those determined by the conventional tests.

Very close agreement was generally found among the values of Young's moduli predicted by the bend tests and those determined by the conventional tests. The largest value of discrepancy of -2.4% occurred for the value of Young's modulus in compression for yield strength level 3 specimens. Very good agreement, then, within 2.4%, was found in the elastic portions of the stress-strain curves predicted by the bend tests and those determined by the conventional tests.

Beyond the elastic region, on the other hand, at least within the 2.0% of total strain to which all specimens were subjected, the flow stress values predicted by the bend tests were generally higher than those determined by the conventional tests. The largest value of discrepancy of +9.8% occurred for the value of flow stress in tension at 2.0% of total strain for yield-strength level 1 specimens. Fair agreement, then, (within 9.8%) was found to exist in the early plastic portions of the stress-strain curves predicted by the bend tests and those determined by the conventional tests.

## IX. SUMMARY AND CONCLUSIONS

- 1. Expressions from Nadai<sup>1</sup>, adapted for experimental exploitation, have been presented that permit the calculation of the early portion of the stress-strain curve, both in tension and in compression, from bend tests. These expressions are applicable to beams, whether or not subjected to pure bending, and are particularly adaptable to materials which fail with relatively small deformations. They are limited to the extent that shear stresses in the bend specimen must be negligible.
- 2. Three of these expressions (two of which are applicable when the stress-strain curve in tension differs from that in compression and one of which is applicable when the stress-strain curves in tension and in compression are equal) have been applied to bend specimens of steel, heat treated to three different yield-strength levels. The bend tests were carried out to strain values of 2.0%.
- 3. Independent tension and compression tests were performed on specimens made from the same stock and subjected to the same heat treatment as the bend specimens. These conventional tests were also carried out to 2.0% strain.
- 4. Stress-strain curves determined from the bend, tension, and compression tests have also been presented. Mechanical properties derived from these curves, including Young's moduli, yield strengths at 0.2% offset, and flow stresses at both 1.0% and 2.0% total strain have also been presented. Maximum percentage scatter and percentage discrepancies have been included with these tabulated values.
- 5. Finally, a comparison of the results predicted by the bend tests and those determined by the tension and compression tests of the specimens, at least within the first 2.0% strain, indicated that:
  - a. Scatter was generally higher in bend tests than in conventional tests, but not objectionable.
  - b. For the present tests, the observation from conventional tests that, at any strain level, the corresponding value of stress in compression was generally higher than that in tension was confirmed in the bend-test results.
  - c. Also for the present tests, the results indicated that the bend test could be regarded as a very good substitute for the conventional tests (within 2.4% agreement) for determining the elastic portions of stress-strain curves and a fair substitute (within 9.8% agreement) for determining the early plastic portions of stress-strain curves.

## ACKNOWLEDGMENTS

The author wishes to acknowledge the invaluable aid provided by Mr. J. I. Bluhm, Chief of Theoretical and Applied Mechanics Research Laboratory and the helpful suggestions offered by Messrs. R. Papirno, F. I. Baratta, and K. Robertson, also of Theoretical and Applied Mechanics Research Laboratory.

#### LITERATURE CITED

1. NADAI, A. *Plasticity*. McGraw-Hill Book Co., Inc., New York, 1931.
2. BLUHM, J. I. *Mechanical Tests - Application of the Bend Test for Determination of the Stress-Strain Curve*. Watertown Arsenal Report No. WAL 110/19, 1950.
3. GROBE, A. H., and ROBERTS, G. A. *The Bend Test for Hardened High Speed Steel*. Transactions of the American Society for Metals, v. 40, 1948.
4. BURGHARDT, A. M., and ELDER, A. S. *Mechanical Tests - Development of a Standard Bend Test for Cemented Carbides*. Watertown Arsenal Report No. WAL 110/16, 1950.
5. DUCKWORTH, W. H. *Precise Tensile Properties of Ceramic Bodies*. Journal of the American Ceramic Society, v. 34, no. 1, 1951.
6. *Standard Methods of Compression Testing of Metallic Materials*. Book of American Society of Testing and Materials Standards, Part 30, 1966.